

[4910-13-U]

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39 [66 FR 14308 3/12/2001]

[Docket No. 2000-CE-61-AD; Amendment 39-12139; AD 2001-05-03]

RIN 2120-AA64

Airworthiness Directives; SOCATA - Groupe AEROSPATIALE Model TBM 700 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD) that applies to certain SOCATA - Groupe AEROSPATIALE (Socata) Model TBM 700 airplanes. This AD requires you to apply Loctite on attaching bolt/screw threads of inboard, central, and outboard carriages; increase tightening torques of associated hardware; and replace central carriage attaching bolts. This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for France. The actions specified by this AD are intended to prevent loose, or the loss of, flap attaching bolts/screws, which could cause rough or irregular control. Such rough or irregular control could lead to the loss of control of the airplane.

**DATES:** This AD becomes effective on April 27, 2001.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulations as of April 27, 2001.

**ADDRESSES:** You may get the service information referenced in this AD from SOCATA Groupe AEROSPATIALE, Customer Support, Aerodrome Tarbes-Ossun-Lourdes, BP 930 - F65009 Tarbes Cedex, France; telephone: (33) (0)5.62.41.73.00; facsimile: (33) (0)5.62.41.76.54; or the Product Support Manager, SOCATA - Groupe AEROSPATIALE, North Perry Airport, 7501 Pembroke Road, Pembroke Pines, Florida 33023; telephone: (954) 894-1160; facsimile: (954) 964-4191. You may examine this information at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2000-CE-61-AD, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4146; facsimile: (816) 329-4090.

**SUPPLEMENTARY INFORMATION:**

**Discussion**

**What events have caused this AD?** The Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, recently notified FAA that an unsafe condition may exist on certain Socata Model TBM 700 airplanes. The DGAC reports two occurrences on Socata model TBM 700 airplanes where, following a flight, a screw of a flap attachment fitting was found partly unscrewed and another was missing. These occurrences are the result of flap vibration.

**What are the consequences if the condition is not corrected?** Loose or the loss of flap attaching bolts/screws could result in rough or irregular control. Such rough or irregular control could lead to loss of control of the airplane.

**Has FAA taken any action to this point?** We issued a proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Socata Model TBM 700 airplanes. This proposal was published in the *Federal Register* as a notice of proposed rulemaking (NPRM) on January 2, 2001 (66 FR 64). The NPRM proposed to require you to apply Loctite on attaching bolt/screw threads of inboard, central, and outboard carriages; increase tightening torques of associated hardware; and replace central carriage attaching bolts.

**Was the public invited to comment?** Interested persons were afforded an opportunity to participate in the making of this amendment. No comments were received on the proposed rule or the FAA's determination of the cost to the public.

Socata has revised Service Bulletin No. SB 70-087 57, dated September 2000 (Amendment 1, dated November 2000), to incorporate minor procedural changes. This AD requires no further action if the original service bulletin is accomplished.

#### **FAA's Determination**

**What is FAA's final determination on this issue?** After careful review of all available information related to the subject presented above, we have determined that air safety and the public interest require the adoption of the rule as proposed except for the addition of the revised service information and minor editorial corrections. We determined that this addition and these minor corrections:

- will not change the meaning of the AD; and
- will not add any additional burden upon the public than was already proposed.

#### **Cost Impact**

**How many airplanes does this AD impact?** We estimate that this AD affects 75 airplanes in the U.S. registry.

**What is the cost impact of this AD on owners/operators of the affected airplanes?** We estimate the following costs to accomplish the modification:

<b>Labor Cost</b>	<b>Parts Cost</b>	<b>Total Cost Per Airplane</b>	<b>Total Cost on U.S. Operators</b>
6 workhours X \$60 per hour = \$360	\$10	\$360 + \$10 = \$370	\$370 X 75 = \$27,750

#### **Regulatory Impact**

**Does this AD impact various entities?** The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

**Does this AD involve a significant rule or regulatory action?** For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption "ADDRESSES".

#### **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### **Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### **PART 39 - AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### **§ 39.13 [Amended]**

2. FAA amends § 39.13 by adding a new AD to read as follows:

# AIRWORTHINESS DIRECTIVE



Aircraft Certification Service  
Washington, DC

U.S. Department  
of Transportation  
**Federal Aviation  
Administration**

*We post ADs on the internet at "av-info.faa.gov"*

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Title 14 of the Code of Federal Regulations (14 CFR) part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety and are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (reference 14 CFR part 39, subpart 39.3).

**2001-05-03 SOCATA - GROUPE AEROSPATIALE:** Amendment 39-12139; Docket No. 2000-CE-61-AD.

(a) What airplanes are affected by this AD? This AD affects Model TBM 700 airplanes, serial numbers 1 through 164, and 166 through 173, that are certificated in any category.

(b) Who must comply with this AD? Anyone who wishes to operate any of the above airplanes must comply with this AD.

(c) What problem does this AD address? The actions specified by this AD are intended to prevent loose, or the loss of, flap attaching bolts/screws, which could cause rough or irregular control. Such rough or irregular control could lead to the loss of control of the airplane.

(d) What actions must I accomplish to address this problem? To address this problem, you must accomplish the following:

Actions	Compliance	Procedures
(1) Apply Loctite on attaching bolt/screw threads of inboard, central, and outboard carriages; increase tightening torques of associated hardware; and replace central carriage attaching bolts with new bolts, part number Z00.N5109337315.	Within the next 25 hours time-in-service (TIS) after April 27, 2001 (the effective date of this AD), if not already done.	In accordance with the ACCOMPLISHMENT INSTRUCTIONS in Socata Service Bulletin SB 70-087, Amendment 1, dated November 2000, and the applicable maintenance manual. No further action is required if accomplished in accordance with Socata Service Bulletin SB 70-087, dated September 2000, and the applicable maintenance manual.
(2) Do not install any central carriage attaching bolts that are not part number Z00.N5109337315 (or FAA-approved equivalent part number).	As of April 27, 2001 (the effective date of this AD).	Not Applicable.

(e) Can I comply with this AD in any other way? You may use an alternative method of compliance or adjust the compliance time if:

- (1) Your alternative method of compliance provides an equivalent level of safety; and

(2) The Manager, Small Airplane Directorate, approves your alternative. Submit your request through an FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

Note 1: This AD applies to each airplane identified in paragraph (a) of this AD, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if you have not eliminated the unsafe condition, specific actions you propose to address it.

(f) Where can I get information about any already-approved alternative methods of compliance? Contact Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4146; facsimile: (816) 329-4090.

(g) What if I need to fly the airplane to another location to comply with this AD? The FAA can issue a special flight permit under sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate your airplane to a location where you can accomplish the requirements of this AD.

(h) Are any service bulletins incorporated into this AD by reference? Actions required by this AD must be done in accordance with Socata Service Bulletin No. SB 70-087 57, Amendment 1, dated November 2000. The Director of the Federal Register approved this incorporation by reference under 5 U.S.C. 552(a) and 1 CFR part 51. You can get copies from SOCATA Groupe AEROSPATIALE, Customer Support, Aerodrome Tarbes-Ossun-Lourdes, BP 930 - F65009 Tarbes Cedex, France; or the Product Support Manager, SOCATA - Groupe AEROSPATIALE, North Perry Airport, 7501 Pembroke Road, Pembroke Pines, Florida 33023. You can look at copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

(i) When does this amendment become effective? This amendment becomes effective on April 27, 2001.

Note 2: The subject of this AD is addressed in French AD 2000-375(A), dated September 20, 2000.

FOR FURTHER INFORMATION CONTACT: Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4146; facsimile: (816) 329-4090.

Issued in Kansas City, Missouri, on February 26, 2001.

Michael Gallagher, Manager, Small Airplane Directorate, Aircraft Certification Service.